**ORIGINAL ARTICLE**



# **Efect of moisture content on the mechanical performance of 3D printed continuous reinforced two‑matrix composite**

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Received: 22 March 2024 / Accepted: 17 June 2024 / Published online: 29 June 2024 © The Author(s) 2024

### **Abstract**

Additive manufacturing, particularly Fused Filament Fabrication, has gained signifcant attraction in recent years. In order to increase the mechanical performances of several components, continuous reinforcements, such as carbon fbers, can be coextruded with a polymeric matrix.

The present study relies on a specifc 3D printing process, called towpreg coextrusion, which exploits continuous carbon fbers covered with an epoxy resin and polyamide (PA) as the thermoplastic matrix, thus obtaining a 3D printed two-matrix composite. Since polyamide is a highly hygroscopic material, the impact of moisture content on the mechanical properties of 3D-printed continuous composites was investigated. Tensile and fexural specimens were manufactured and tested under both undried and dried conditions. Drying treatment was carried out at a temperature of 70 °C for 2 h in oven, with weight measurements before and after for quantifying weight loss and then the moisture removal. Additionally, through thermogravimetric analysis, the thermal stability of the material was assessed. It was observed that the drying process allows for a reduction of up to 0.56% by weight of moisture in the specimens. Thus, the drying process led to an improvement in the mechanical properties of the material. Specifcally, the tests reveal a 15% increase in tensile strength and an 11.5% increase in fexural strength following the drying process, reaching values of 392.78 MPa and 151.06 MPa, respectively. Similarly, an increase in the tensile and fexural moduli was noted in the treated specimens. Finally, fractured samples underwent optical and scanning electron microscopy analysis, through which diferent fracture mechanisms of the material and the presence of macrovoids and microvoids attributable to the 3D printing process were observed. Knowledge of deposition defects represents an important starting point for the improvement of the process and the mechanical properties obtained to date. This research provides valuable insights into optimizing 3D-printed continuous composites, emphasizing the importance of moisture control for superior mechanical performance in industrial applications.

**Keywords** Additive manufacturing · Coextrusion process · Two-matrix composite materials · Moisture efect · CFRP · TGA

#### **Highlights**

- •Tensile and fexural specimens were printed using a 3D printing process with continuous carbon-epoxy flament (CCF) and a nylon-based material as the thermoplastic matrix.
- •The impact of moisture content on the mechanical properties of 3D-printed continuous composites was investigated.
- •Drying treatment was carried out.
- •Thermogravimetric and scanning electron microscopy analyses have been performed.
- •The drying process improves mechanical properties.

Marina Andreozzi and Serena Gentili made equal contribution.

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## **1 Introduction**

Additive manufacturing has shown remarkable progress in recent years; in particular, Fused Filament Fabrication (FFF) has emerged as the leading technique in 3D printing technologies  $[1, 2]$  $[1, 2]$  $[1, 2]$  $[1, 2]$ . FFF, known for its versatility and cost-effectiveness, has received widespread attention for its applicability in various felds. One of the main limitations of this technology is the achievable mechanical performance. Indeed, even if toughened thermoplastics or techno-polymers are used (such as PA or PEEK), it is difficult to reach tensile strength higher than 100 MPa [\[3](#page-8-2)]. A method to reinforce 3D printing flament is to produce a composite by adding short

carbon fbers [[4\]](#page-8-3). In this way, it is possible to increase the performances of 3D-printed components, also reaching tensile strength higher than 150 MPa. However, these values are not sufficient for different industrial applications where high performances are mandatory. Another possibility to obtain much higher performances is to manufacture a 3D-printed composite with continuous fibers  $[5, 6]$  $[5, 6]$  $[5, 6]$  $[5, 6]$ . The study conducted by Dul et al. [[7](#page-8-6)] found that specimens printed in PA with continuous carbon fber achieved 34% higher tensile performance in strength and 147% higher stifness than pure PA. In bending, they also found an increase in strength and stifness of 29% and 140%, respectively.

The use of continuous fbers in these technologies has received significant attention, with the development of numerous successful prototypes and commercial printers [\[8](#page-9-0)–[10\]](#page-9-1).

Given the growing interest in the 3D printing of continuous fber composites, a signifcant increase in the employed technologies has been reported in recent years [[11](#page-9-2)]. Currently, the range of printable materials, the size of produced parts, achievable production volumes, and application felds have expanded. As of today, a formal classifcation of longfber composite printing techniques does not exist. However, it is possible to categorize the processes based on how the fber and matrix are directed to the printing nozzle and how they are deposited [\[12](#page-9-3), [13](#page-9-4)].

In the in-situ impregnation technologies, dry fbers are pushed through the nozzle, in which they are impregnated with the matrix, thus realizing the coextrusion process. Two other existing technologies are in-line impregnation and insitu consolidation. In the former, the fber is impregnated while being pushed through the nozzle. In the latter, however, a thermoplastic flament (towpreg) is consolidated in situ by a pressure roller during deposition, while an external heat source facilitates polymerization. When the preimpregnated flament is directly heated and extruded trough the nozzle, the relative manufacturing technology is named towpreg extrusion $[14]$  $[14]$ . Another technique is the coextrusion of towpreg. In this case, instead of dry fber, a thin towpreg/preimpregnated flament is used, heated, and coextruded together with the matrix. If the towpreg matrix is the same as the coextrusion matrix, a monomatrix composite is obtained and, if the matrix flament material is a thermoplastic, the approach is referred to as continuous fber-reinforced thermoplastic composites [\[10](#page-9-1), [15](#page-9-6)]. On the contrary, a bimatrix composite is referred to when the resin used for tow fabrication difers from the coextrusion resin [\[16–](#page-9-7)[18\]](#page-9-8). An example is the coextrusion of thermoplastic resin and preimpregnated thermosetting resin carried out by the Composer 3D printer developed by Anisoprint. This system exploits a 1.5-K continuous carbon fber flament impregnated with an epoxy resin and PA as the main thermoplastic matrix. The carbon fber flament is reinforced with the thermoset

resin through a patented binderization process in order to increase its processability in the 3D printer. The binderization process aims to provide more stifness to the flament, thus avoiding the fufng of the fbers during the movements. However, once the fber bundle crosses the heated nozzle, the thermoset resin is heated at a temperature higher than its glass transition temperature, thus allowing an easy deposition of the fbers also in curved geometries. Furthermore, in the heated nozzle , the impregnation of the fber bundle with the thermoplastic polyamide occurs. This phase is crucial to obtain high-quality components without defects such as voids. In fact, in the case of 3D-printed composite structures, it is very important to explore the surface topology, cross-sectional view, and nature of the fracture to understand whether structural defects such as gaps and poor interlaminar adhesion are present [[11](#page-9-2)]. Indeed, the presence of voids leads to a decrease in mechanical performances as they act as nucleation points for cracks. A parametric study of a type of continuous two-matrix carbon fber composite constituted by PET-G and prepreg carbon fber bundles, using coextrusion of towpreg technology [[16\]](#page-9-7), showed an increase in maximum fexural strength and modulus of 74% and 93% over those of pure PETG parts, and the minimum porosity content of 3.19% is achieved. The presence of voids within the molded CF/PA6 continuous composite was found to have a substantial negative impact on mechanical performance, as transverse tensile strength and fexural strength increased by 78% and 93%, respectively, with the void content decreasing from 12 to 6% [[19\]](#page-9-9). In addition, voids typically increase moisture absorption, another aspect that can lead to a reduction in performances. Moreover, PA is known for its high hygroscopicity, which predisposes it to absorb moisture during the 3D printing process [[20,](#page-9-10) [21](#page-9-11)]. No scientifc research can be found concerning the efect of moisture absorbed by the polyamide material on the mechanical performances of 3D-printed continuous composites.

In this framework, the present study aims to address a critical knowledge gap by investigating the efect of moisture content on the mechanical performances of 3D-printed continuous fber composites fabricated using the towpreg coextrusion process. In this context, no prior research has explored this relationship. Thus, this study will provide novel insights in relation to how moisture absorption in the polyamide matrix infuences the mechanical properties of 3D-printed structures.

The aim of this research was fulflled by conducting tensile and fexural characterization on specimens manufactured with the towpreg coextrusion. Specimens will be tested in both undried and dried conditions (moisture removed through heating at 70  $\degree$ C for 2 h). The thermogravimetric analysis will be conducted to assess the thermal stability of the materials, and the amount of moisture removed will be quantifed by weighing samples before and after drying.

Undried samples will be tested immediately after 3D printing. Finally, fractured samples will be analyzed using optical and scanning electron microscopy (SEM) to investigate the presence and infuence of voids on mechanical performance.

## **2 Materials and methods**

#### **2.1 3D printer and material**

The machine used for the 3D printing of continuous fberreinforced composites was the Composer A3 printer developed by Anisoprint Inc. This system exploits the additive manufacturing process illustrated schematically in Fig. [1.](#page-2-0) This process allows to 3D print a composite material consisting of continuous carbon fbers and a thermoplastic matrix. Thus, inside the heated chamber, the thermoplastic material is melted and extruded simultaneously with the continuous flament through the same extrusion nozzle. For this reason, the process is called composite fber coextrusion (CFC). The utilization of dedicated cutting equipment enables the segmentation of continuous fbers, thus facilitating the fabrication of geometrically complex and dimensionally precise structures.

As far as the continuous carbon-epoxy flament (CCF) is concerned, it was supplied by the machine manufacturer and consists in a tow of 1.5-K carbon fbers with an average diameter of 7 μm. In order to increase the stifness and the processability of the CCF, the tow is covered, through a binderization process, with an epoxy-based thermoset resin. This CCF flament has remarkable mechanical properties, including an elastic modulus of 150 GPa, a tensile strength of 2200 MPa, and a carbon fber volume fraction



<span id="page-2-0"></span>**Fig. 1** Scheme of CFC process

of 60%, as reported in the technical datasheet provided by the manufacturer.

Concerning the thermoplastic matrix, the Anisoprint CFC process exploits a specially developed nylon-based material from Polymaker. This resin is characterized by a tensile and fexural strength of 57 MPa and 69 MPa, respectively; an elastic modulus in tension and fexure of 1440 MPa and 1580 MPa, respectively; and an elongation at break of 15.86%. In addition, this resin exhibits a low-viscosity profle, enhancing the interlayer adhesion of the fbers. Moreover, its quick cooling and solidifcation properties contribute signifcantly to the precision of fber placement, ensuring enhanced quality in the fnal component. This resin is characterized by a low absorption of moisture.

The 3D printing process started with the design of the CAD geometries using dedicated software. Then, the generated mesh fles were imported into the proprietary slicing software (Aura) that converted the 3D models into instructions for the 3D printer, generating a GCODE fle. Aura software allows the confguration of various printing parameters, including infll, fow multiplier, reinforced infll pattern, guide direction, printing speed, and printing temperature (Table [1](#page-2-1)).

The printing parameters, as recommended by the Anisoprint experts, were chosen to optimize the mechanical properties of the 3D-printed specimens. Additionally, the implementation of a solid-reinforced infll pattern ensures a 100% infll with continuous carbon fber, signifcantly enhancing the characteristics of the components.

#### **2.2 Thermogravimetric analysis**

The thermal stability of the preimpregnated fber bundle and composites (5–10 mg) was investigated through thermogravimetric analysis (TGA), using TA Instruments, TGA 55 equipment, at a heating rate of 10 °C min−1 from 40 to 900 °C in a nitrogen gas atmosphere (100 mL min<sup>-1</sup>). The composite specimens, before being submitted to this analysis, were stored at two diferent conditions, to examine the efect of humidity absorption: some of them were stored at room temperature (RT) and maintained at room humidity (undried); the remaining specimens were dried in an oven at 70 °C for about 2 h (dried). The drying treatment was

<span id="page-2-1"></span>**Table 1** Printing parameters

Average printing speed	$6 \text{ mm/s}$
Reinforced infill pattern	Solid
Macrolayer height	$0.32$ mm
Extrusion width	$0.65$ mm
Extruder temperature	250 °C
Build plate temperature	60 °C

applied also for the PA. For TGA analyses, samples of these diferent composite specimens and those of PA were used.

## **2.3 Drying procedure**

To analyze the effect of moisture on the properties of 3D-printed composites, mechanical tests were conducted on both dried and undried specimens. The drying process, aimed at moisture removal, involved a specifc exsiccation method. It consists of placing the 3D-printed samples in a ventilated oven at a temperature of 70  $\degree$ C for 2 h. The effectiveness of this drying procedure was quantitatively assessed by measuring the weight of the samples before and after drying using an analytical balance with a resolution of 0.005 g.

## **2.4 Tensile test**

To assess the mechanical performances of the 3D-printed composites, tensile tests were carried out. The test specimens were fabricated in accordance with ASTM D3039 standards: 2 mm in thickness, 15 mm in width, and 250 mm in length. To prevent potential defects due to the curving of continuous fbers at the specimen ends, an ellipticalshaped geometry was adopted (as depicted in Fig. [2\)](#page-3-0) and the tensile specimens were cut from the rectilinear part of the 3D-printed sample. As can be seen from the fgure, the fbers were oriented in the direction of the tensile load, thus allowing for the highest achievable tensile properties.

To guarantee repeatability, diferent tensile specimens were printed and tested. More in detail, 4 samples were printed and tested immediately after printing while the other 4 samples were subjected to the drying process reported in Sect. 2.4 before testing. This allows us to investigate the efect of moisture on the tensile properties of the 3D-printed composites. The tensile tests were conducted using a Zwick/ Roell Z050 machine with a constant test speed of 2 mm/min. During the testing process, a 50-kN load cell was utilized to measure the applied load, while an optical extensometer was employed to record the nominal strain along the loading direction, by which the maximum value of tensile strength was calculated.

## **2.5 Flexural test**

The flexural behavior of the 3D-printed composite was characterized through a three-point bending test. The test was conducted following the ASTM D7264 standard, and the specimen dimensions were 128 mm in length, 4 mm in thickness, and 15 mm in depth. The span-to-thickness ratio was 32:1.

In this case, specimens were fabricated using a rectangular geometry, sized in accordance with the standard, with an initial fber protrusion in order to reduce the risk of fber under-extrusion that can lead to missing fbers in the specimen during the initial printing phase.

The extruded part visible at the sample corner indicates the onset of the 3D printing process, a technique employed to ensure uniform fber distribution throughout the specimen. After the printing process and prior to mechanical testing, these protrusions are dedicated tools.

A universal testing machine equipped with a 25-kN load cell was used to perform the tests. According to the standard, a loading nose and supports of 5-mm radii were employed, and the crosshead speed was set to 2 mm/min. During the test, the load and displacement (with contact microestensimeter) were recorded. The stress and the strain for any point on the load-defection curve were calculated following the ASTM D7264, Procedure A. The fexural modulus was calculated as the slope of the stress-strain curve taken in the strain range between 0.001 and 0.003.



<span id="page-3-0"></span>**Fig. 2 a**) Tensile specimen process. **b**) Areas of the elliptical shape to extract the tensile specimens. **c**) Scheme of the carbon fber composite deposition

Also, in this case, a total of 8 specimens were 3D printed. Four samples were tested immediately after printing, while the other 4 were subjected to the moisture removal procedure reported in Sect. 2.4.

#### **2.6 Optical and scanning electron microscopies**

The morphological analysis of the fracture surfaces from specimens subjected to flexural and tensile tests was conducted using the Leica DMi8 optical microscope (Leica Microsystems GmbH, Wetzlar, Germany). Additionally, for more detailed inspection, the SEM EVO 10 was employed to achieve high-resolution imaging of the fractured surfaces of the tested specimens. Prior to the SEM observations, the specimens underwent a gold metallization process to enhance conductivity and facilitate the acquisition of highquality electron microscope images, then observed using an accelerating voltage of 20 kV.

#### **2.7 Statistical analysis**

Mechanical measurements were reported as mean  $\pm$  standard deviation (mean  $\pm$  SD). One-way ANOVA test was used for statistical analysis using the Statgraphics Plus 5.1. Program (Manugistics Corp., Rockville, MD). To diferentiate samples, Fisher's least signifcant diference (LSD) was used at the 95% confdence level. Diferences were considered statistically significant for  $p < 0.05$ .

## **3 Results and discussion**

#### **3.1 Thermogravimetric analysis**

Figure [3](#page-5-0) shows the residual mass and the corresponding derivative of a sample of the impregnated fiber bundle (Fig.  $3(a)$  $3(a)$ ); these curves are reported also in Fig.  $3(b)$  and (c) together with those of PA and undried and dried composites. The initial weight loss of preimpregnated fiber bundle starts from 230 °C (symbolized as  $T_0$  in Fig. [3](#page-5-0)(a)), which is related to the initial thermal decomposition behavior of the epoxy resin. It follows with a 5 wt% weight loss ( $T_{onset}$  fiber bundle) at 365 °C according to the literature  $[16]$  $[16]$  $[16]$ . The thermal decomposition is due to the cleavage of aromatic groups in the epoxy matrix, and the main degradation peak is centered at 415 °C. At 600 °C, a reduction in material weight of about [3](#page-5-0)0% has been calculated (Fig.  $3(a)$ ). Therefore, the weight fraction of carbon fiber is close to 70% (Fig. [3](#page-5-0)(a)).

TGA measurements of the fber bundle, thermoplastic polymer, and dried and undried composites are displayed in Fig.  $3(b)$ , (c). The TG (Fig.  $3(b)$ ) and DTG (Fig.  $3(c)$ ) curves of both composites show similar thermo-degradative behavior. The presence of multi-degradation steps in composite

specimens indicates the presence of diferent components characterized by diferent temperatures of decomposition. The frst residual mass step of the undried composite sample (see inset in Fig.  $3(b)$ ) is due to the presence of moisture that can be removed at a temperature below 130 °C. The corresponding weight loss is estimated at around 0.50% (residual mass loss is close to 99.5%). This result is in agreement with those of Tables [2](#page-6-0) and [3,](#page-6-1) thus confrming the presence of a small amount of water in the undried composites, which infuences their mechanical properties, as reported above.

The main peak of DTG curves corresponds to a mass loss at 454 °C (Fig.  $3(c)$  $3(c)$ ), as a result of the thermal degradation behavior of PA, clearly present also in the composites [\[22](#page-9-12)]. In addition, in the composite curves, also a shoulder can be observed at 384 °C. This value is a little bit less than that of the DTG peak at 415 °C of the stand-alone epoxy resin in the fber bundle, suggesting that the degradation process of this resin is made more complicated by the contact with the PA matrix.

Finally, at 900 °C, the residual mass of PA was estimated at 1.7%, while for composites 46.6% and 45.8% for undried and dried (Fig.  $3(c)$  $3(c)$ ), respectively. This last variation can be attributed to the diferent polymer content in the samples analyzed by TGA, as a consequence of slightly diferent amounts of polymers deposited on carbon fbers during the 3D printing process, according to the SEM analysis (Fig. [7](#page-7-0)).

#### **3.2 Tensile tests**

Figure [4](#page-6-2) shows the typical nominal stress vs. nominal strain curves obtained by performing tensile tests on undried and dried specimens.

In Table [2](#page-6-0), the average value of maximum tensile strength  $(\sigma_{\text{tmax-avg}})$ , tensile strain at the peak stress point  $(\varepsilon_{\text{tmax-avg}})$ , Young's modulus  $(E_{t_{\text{avg}}})$ , and the respective standard deviations are reported. Furthermore, the average weight loss related to moisture removal is also reported.

From the results, it can be observed that the drying process led to an average weight reduction of 0.5% and an enhancement in mechanical properties. Indeed, the tensile stress of dried materials exhibited a 15% increase while no significant differences ( $p < 0.05$ ) of tensile Young's modulus and tensile deformation at break were registered.

The increase of both tensile strength and elastic modulus after the drying process is confrmed by literature. Polyamide is well known for its hygroscopic nature and the moisture absorption results in a hydrolytic degradation of the polymer, particularly at the interface with the reinforcing fbers [\[21](#page-9-11), [23](#page-9-13), [24\]](#page-9-14). The drying process minimizes this degradation, preserving the structural integrity of the polymer and the fber-matrix interface. This preservation is fundamental to guarantee an efective stress transfer mechanism from the <span id="page-5-0"></span>**Fig. 3** Thermogravimetric curves of an impregnated fber bundle (**a**), TG and DTG curves, residual mass (**b**). and derivative mass loss (**c**) curves of the fber bundle, PA, and composites in the nitrogen atmosphere



<span id="page-6-0"></span>**Table 2** Mechanical properties of dried and undried tensile specimens

	$\sigma_{\text{tmax\_avg}}$ (MPa) $\mathcal{E}_{\text{tmax\_avg}}$ (%) $E_{\text{t\_avg}}$ (GPa)		Average weight loss (% )
	Undried $340.49 \pm 8.92^a$ $0.82 \pm 0.01^a$ $41.13 \pm 1.13^a$ -		
Dried	$392.78 +$ 11.39 <sup>b</sup>	$0.93 \pm 0.02^{\text{a}}$ 43.33 $\pm 1.01^{\text{a}}$ 0.51 $\pm 0.02$	

Diferent superscripts (a and b) within the same column indicate significant differences among formulations ( $p < 0.05$ )

<span id="page-6-1"></span>**Table 3** Mechanical properties of dried and undried fexural specimens

$\sigma_{\text{fmax\_avg}}$ (MPa) $\mathcal{E}_{\text{fmax\_avg}}$ (%) $E_{\text{f\_avg}}$ (GPa)		Average weight loss (% )
Undried $133.74 \pm 2.65^a$ $0.56 \pm 0.01^a$ $31.77 \pm 0.39^a$ -		
Dried $151.06 \pm 4.01^b$ $0.59 \pm 0.05^a$ $30.48 \pm 0.83^a$ $0.56 \pm 0.07$		

Diferent superscripts (a and b) within the same column indicate significant differences among formulations ( $p < 0.05$ )



<span id="page-6-2"></span>**Fig. 4** Average tensile curves of dried and undried composite specimens

matrix to the fbers and, thus, to ensure the high mechanical properties of the examined composite material.

#### **3.3 Flexural tests**

A representative curve for both dried and undried specimens is displayed in Fig. [5](#page-6-3), whereas the corresponding characteristic parameters are reported in Table [3.](#page-6-1)

The dried specimens exhibit an increase of the maximum flexural resistance ( $\sigma_{\text{tmax} \text{avg}}$ ) of 11.5% (significant differences ( $p < 0.05$ ) after the post-drying process) while no particular variation was registered for fexural Young's



<span id="page-6-3"></span>**Fig. 5** Average fexural curves of dried and undried composite specimens

modulus  $(E_{t_{avg}})$  and of the flexural strain at the peak stress point ( $\epsilon_{\text{max\_avg}}$ ). The table also details the standard deviation for each parameter and the average weight loss relative to the drying process.

Similar results observed in tensile tests are also refected in fexural tests, emphasizing the comprehensive enhancement of the material's mechanical properties. The drying process, by mitigating moisture-induced degradation, not only increases the material's resistance to tensile loads but also signifcantly enhances its ability to withstand fexural stresses. The latter increase further validates the crucial role of the drying process in preserving the structural integrity of the polymer and the fber-matrix interface.

## **3.4 Optical and scanning electron microscope analyses**

Figure [6](#page-7-1) shows the optical micrography at  $8 \times$  of a tested fexural specimen. A non-homogenous fracture surface can be seen, thus suggesting that multiple fracture mechanisms occurred, including matrix cracking, fber-matrix debonding, and fber breakage. This behavior can be caused by deposition defects and non-uniform impregnation of the fber during the coextrusion process.

The presence of a non-homogenous fracture surface indicates the complexity of the failure process in the composite material. Such behavior can be attributed to several factors. The frst one is matrix cracking**,** namely, the initial failure mechanism in composite materials under fexural loading. Matrix cracking can occur due to the brittle nature of the matrix material, which may not be able to withstand the tensile stresses induced during bending. These cracks can propagate, leading to further weakening of the composite structure. The second factor is fber-matrix debonding**.** As

<span id="page-7-1"></span>**Fig. 6** Fractured surface 8× optical micrograph of fexural specimens



the matrix cracks, the load is transferred to the fbers. If the bonding between the fbers and the matrix is weak, debonding can occur. This debonding reduces the load transfer efficiency and contributes to the overall failure of the composite. The debonding observed in the micrograph suggests that there may have been issues with the fber-matrix interface, possibly due to insufficient adhesion or improper curing processes. Lastly, the non-homogenous fracture surface is related to fber breakage**.** Once the fbers are bearing the majority of the load, they may eventually reach their tensile strength limit and break. Fiber breakage is a critical failure mechanism as it signifes the ultimate load-bearing capacity of the composite. The observation of fber breakage in the micrograph indicates that the fbers were subjected to high stress levels, which they could not withstand indefnitely.

The occurrence of these multiple fracture mechanisms can be linked to manufacturing imperfections. Specifcally, deposition defects and non-uniform impregnation of the fbers during the coextrusion process are likely contributors. Deposition defects might include voids, uneven distribution of the resin, or contaminants that weaken the material structure. Non-uniform impregnation refers to the inconsistent distribution of the matrix material around the fbers, leading to areas with poor bonding and increased susceptibility to cracking and debonding.

Figure [7](#page-7-0) shows the cross-section SEM images of a tensile specimen. In Fig.  $7(a)$  $7(a)$ , the fiber bundle can be observed, along with the individual carbon fbers that constitute the fber flament. The structure shows discontinuities between deposited bundles and the presence of macrovoids. Figure [7](#page-7-0)(b) is a higher magnifcation image and puts the evidence of the presence of microvoids between the single carbon fber flaments.

The occurrence of both macrovoids and microvoids within the composite structure can be attributed to the 3D printing process. Specifcally, macrovoids arise from defects during flament deposition, which results in unflled spaces between adjacent extruded parts. These voids can form due to inconsistencies in the extrusion process, such as variations in temperature, pressure, or nozzle movement, leading to incomplete bonding between layers. On the other hand, microvoids, which are present within the flament bundles, indicate suboptimal impregnation during the extrusion phase. This suggests that the resin did not fully penetrate the

<span id="page-7-0"></span>**Fig. 7 a**)  $200 \times$  and **b**)  $1000 \times$ SEM magnifcations of 3D-printed composite



fiber bundles, possibly due to insufficient wetting or uneven distribution of the resin. These microvoids act as stress concentrators and signifcantly reduce the mechanical properties of the 3D-printed composite by creating weak points that can initiate cracks and propagate under load.

The presence of both macrovoids and microvoids can compromise the structural integrity and mechanical performance of the composite material, leading to reduced tensile strength and durability. Addressing these defects through process optimization, such as improving flament deposition techniques and ensuring better impregnation, is crucial for enhancing the quality and performance of 3D-printed composites. Future work should focus on refning the 3D printing parameters and incorporating real-time monitoring systems to detect and mitigate the formation of voids during the printing process.

## **4 Conclusion**

In this study, the fabrication of a composite material for FDM 3D printing was investigated, with a particular emphasis on examining how moisture absorption afects its tensile and fexural properties. This 3D-printed composite material is engineered using continuous carbon fbers that are bonded with an epoxy resin and further impregnated with polyamide, thereby creating a dual-matrix composite structure. The TGA measurements permitted to determine the thermo-degradative profle and highlighted the diferent content of polymeric deposition on carbon fbers during the 3D printing process. The analysis revealed that the polyamide used in the 3D printing process, also if it is defned by the manufacturer with low hygroscopicity, is subjected to moisture absorption. However, using a drying process allows to remove up to 0.56% of this moisture by weight from the samples, leading to enhanced tensile and fexural properties, as evidenced by increased strength and stifness. Furthermore, SEM analysis highlighted the presence of both microvoids and macrovoids in the 3D-printed composites, which not only led to a reduction of structural integrity but also potentially increased moisture absorption during the service phase. Moreover, it is critical to address the void content within the composites, as their presence can compromise the component's strength and amplify moisture absorption. For the latter, surface treatments such as spray and liquid coatings can be viable strategies to waterproof the surface and enhance the material's durability.

Future works will concentrate on optimizing 3D printing parameters to further refne the mechanical performance of these composites. More detailed studies will be conducted towards reducing void content and limiting moisture absorption in order to signifcantly elevate the quality and resilience of 3D-printed polyamide composites.

**Acknowledgments** The authors gratefully acknowledge the Marlic laboratory for the SEM analysis and Dr. Giuseppe Pandarese for his support in the experimental procedures.

**Author contribution** Andreozzi marina: writing, formal analysis, data curation, investigation. Serena Gentili: writing, formal analysis, data curation, investigation. Pietro Forcellese: supervision. Tiziano Bellezze: writing — reviewing and editing, supervision. Valeria Corinaldesi: supervision. Francesca Luzi: writing, formal analysis, data curation, investigation. Alessio Vita: methodology, investigation, writing—reviewing and editing, supervision.

**Funding** Open access funding provided by Università Politecnica delle Marche within the CRUI-CARE Agreement.

#### **Declarations**

**Conflict of interest** The authors declare no competing interests.

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