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(Article begins on next page)

1     **ANALYSIS OF SHEAR-TORQUE FATIGUE TEST FOR BITUMINOUS PAVEMENT**  
2    **INTERLAYERS**

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10    **ABSTRACT**

11    *The laboratory characterisation of bituminous interlayers is a challenging issue because of the*  
12    *difficulty of establishing loading conditions that can simulate field behaviour. This paper*  
13    *describes a shear-torque fatigue test procedure aimed at investigating the fatigue behaviour of*  
14    *bituminous pavement interlayers. A preliminary experimental investigation, able to evaluate the*  
15    *damage process occurring in both single- and double-layered specimens, showed that the*  
16    *presence of the interface constitutes a weakness zone, noticeably influencing the shear fatigue*  
17    *performance of the pavement structure. Thus, a model for the description of the evolution of the*  
18    *interlayer damage by using the variation of the interlayer complex shear modulus was proposed*  
19    *and preliminarily validated. Moreover, a new fatigue failure criterion has been proposed to be*  
20    *used in the result analysis of the shear-torque fatigue tests.*

21    **Keywords:** asphalt pavement; bituminous interlayer bonding; cyclic loading; shear-torque  
22    fatigue test; damage accumulation.

23

24 **1. Introduction**

25 It is widely recognised that interlayer bonding conditions between bituminous layers directly  
26 influence the overall pavement performance. Indeed, a proper interlayer bonding is fundamental  
27 for transferring normal and shear stresses induced by traffic loading throughout bituminous  
28 layers, providing to the pavement structure the uppermost load-bearing capacity. On the contrary,  
29 poor bonding between bituminous layers may cause low performance, which in turns could lead  
30 to slippage cracking, top-down cracking and permanent deformations [1-4], especially in the areas  
31 where the vehicles frequently accelerate, brake or turn sharply [5]. It is well known that interlayer  
32 bonding is related to a large number of factors, such as type of materials (e.g., mixture type, binder  
33 type, aggregate texture), construction characteristics (e.g., tack-coat, compaction/voids, surface  
34 roughness, contamination, reinforcement) and loading/environmental aspects (e.g., temperature,  
35 traffic speed, ageing process) [5-11] that should be taken into account during the design process.

36 Notwithstanding the fundamental importance of interlayer bonding condition, the  
37 pavement design methods currently lack a standardised approach to characterise interlayers. One  
38 of the main reasons is that the laboratory characterisation of bituminous interlayers is a  
39 challenging issue because of the difficulty of establishing loading conditions that can simulate  
40 field behaviour. During the last decades, several laboratory equipment has been developed to  
41 investigate the interlayer bonding based on different testing methods, loading conditions as well  
42 as specimen geometry and preparation. These tests can be classified into four main groups: tensile  
43 (pull-off), torque, wedge splitting and direct shear tests (with and without the application of  
44 normal load) [5,6,12]. Depending on the test device, these testing configurations simulate the  
45 mode of interlayer bonding failure (e.g., failure due to tensile or shear stresses) [5,6,12]. Among  
46 all, shear testing devices are the most popular test configuration for laboratory assessment of  
47 bonding condition and allow, in some cases, also the application of normal load on the interface  
48 plane to better simulate traffic loading. Typical testing protocol adopts monotonic condition in  
49 strain-controlled mode, by applying a constant displacement rate between two pavement layers  
50 until failure. It allows the determination of the interlayer shear strength (ISS), which represents

51 the maximum shear stress the interface can withstand before failure. Recent studies confirmed the  
52 validity of the time-temperature superposition principle for the ISS, highlighting the viscoelastic  
53 nature of such failure parameter through the adoption of sigmoidal master curves [13,14].  
54 Notwithstanding, such a procedure entails substantial simplification of the real pavement  
55 behaviour and the ISS can only be used for quality control purposes as it does not provide any  
56 information on bonding deterioration during the pavement lifespan except in those cases where  
57 the shear stress induced by traffic is very high (i.e., where the vehicles frequently accelerate, brake  
58 or turn sharply). Thus, since the traffic loadings applied to the pavement surface are repetitive and  
59 induce progressive fatigue damage and/or accumulation of permanent deformations, it is well  
60 recognised that a quasi-static monotonic load is not appropriate to correctly simulate the traffic  
61 effect that the pavement structure undergoes during its lifetime.

62 This drawback can be solved by using cyclic bond tests, consisting in the repetitive  
63 application of a load or a displacement between two pavement layers, as they are expected to be  
64 able to better estimate the load-induced damage on the interlayer zone over the pavement lifetime.  
65 It has been shown [15-23] that cyclic bond testing methods can be effectively used to investigate  
66 the interlayer pavement properties even though, compared with the monotonic condition, the  
67 corresponding devices are generally more sophisticated, as well as the tests are time-consuming  
68 and require a considerable computational effort for the result analysis. Therefore, although cyclic  
69 bond tests are more effective in simulating the real mechanism experienced in-situ, very few bond  
70 test devices have been used in cyclic conditions so far. The cyclic bond tests currently available  
71 differ in terms of the working scheme, loading method (stress- or strain-controlled), testing  
72 conditions and failure criterion adopted to evaluate the fatigue life of the interlayer [15-23].  
73 However, it is worth noting that the chosen control mode and testing conditions can have an  
74 important impact on the testing results.

75 In this framework, the Task Group 3 “Pavement multilayer system” of the RILEM TC  
76 272-PIM “Phase and Interphase behaviour of bituminous Materials” carried out an interlaboratory  
77 test [15], which aimed at defining suitable test methods and deepening the understanding of the

78 pavement interlayer shear behaviour in terms of stiffness properties (small-strain thermo-  
79 viscoelasticity) and fatigue properties (accumulated damage).

80           Within this interlaboratory test, one of the participating laboratories [23] developed a new  
81 testing method, namely shear-torque fatigue test, to carry out fatigue tests on double-layered  
82 cylindrical specimens. This test allows to sophisticatedly investigate the fatigue behaviour of  
83 interlayers by applying a sinusoidal torsional load across the interface and monitor the resulting  
84 angular rotation, even if a non-uniform (i.e., linear) shear stresses distribution at the interface is  
85 obtained. In a recent study, Ragni et al. [23] showed that, during shear-torque fatigue tests, the  
86 damage (i.e., the highest permanent deformations) is mainly concentrated in the interlayer region  
87 of double-layered specimens. Therefore, it can be assumed that, for the most part, the specimen  
88 failure is due to the interface damage, which could be modelled by using the variation of the  
89 interlayer complex shear modulus.

## 90 **2. Objectives**

91 This paper presents the modelling of the behaviour of pavement bituminous interlayers during  
92 shear-torque fatigue tests, by including a preliminary validation of the model and a new fatigue  
93 failure criterion. To these purposes, three main phases have been considered as follows:

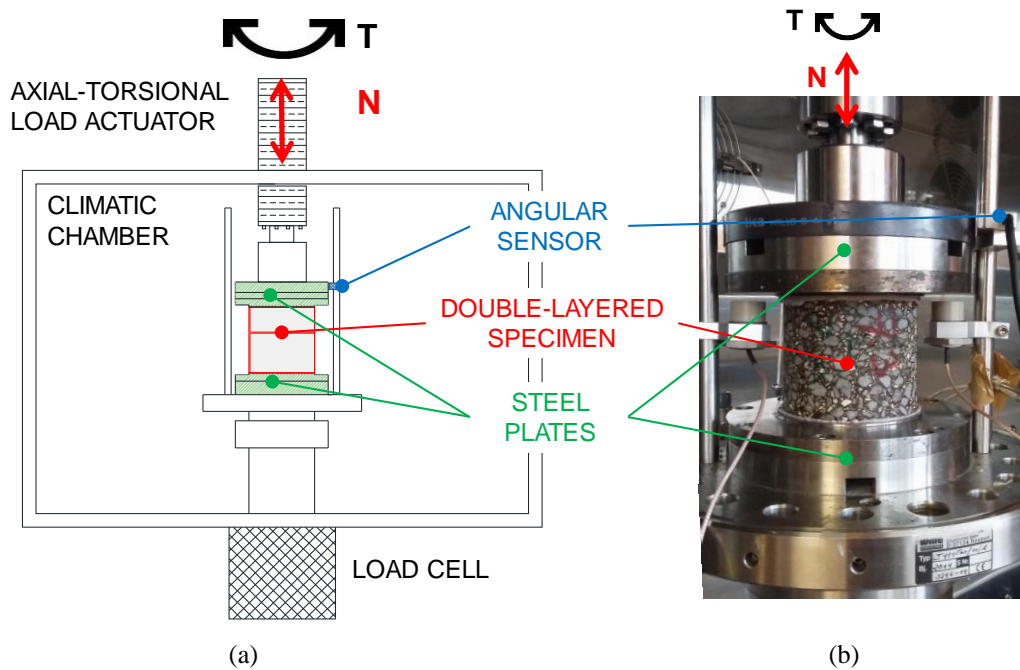
- 94       - after the description of the shear-torque fatigue test configuration, the equations for  
95       describing the material behaviour and the model of the evolution of the interlayer  
96       complex shear modulus are presented;
- 97       - shear-torque fatigue tests performed on both single- and double-layered bituminous  
98       specimens have been used for preliminary validation of the model;
- 99       - a new fatigue failure criterion has been proposed for the result analysis of the shear-torque  
100       fatigue tests, allowing the determination of the interlayer shear fatigue curve.

## 101 **3. Shear-torque fatigue test procedure**

102 In all the three phases of the investigation, a shear-torque fatigue device supplied by a servo-  
103 hydraulic testing equipment, available at the Laboratoire GC2D of the University of Limoges, has

104 been used to carry out the experimental program. The testing device, able to apply a torque of  
105  $\pm 1,000$  Nm and an axial load of  $\pm 100$  kN, is equipped with a monitoring control system,  
106 associated with a temperature-controlled chamber.

107 Tests were carried out in load-controlled mode at a frequency of 10 Hz and a temperature  
108 of 20 °C. A sinusoidal torque with an amplitude  $T_0$  and a constant axial compression load  $N$  of  
109 0.05 kN were applied to the cylindrical specimen. A general view of the working scheme of the  
110 shear-torque fatigue test is shown in Fig. 1. In order to measure the sinusoidal torsional rotation  
111 angle, a non-contact angular sensor located on the upper steel plate was used (accuracy of 0.001°)  
112 (Fig. 1).

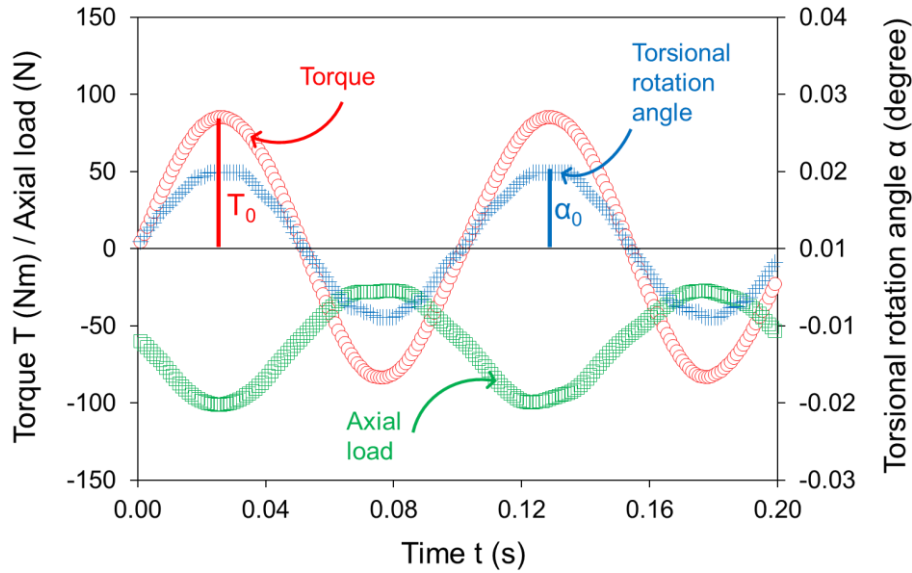


113 (a) (b)  
114 Fig. 1. Shear-torque fatigue test set-up: (a) schematic view and (b) picture in the temperature-  
115 controlled chamber.

116 Prior to run the fatigue test, each specimen was glued with an epoxy resin to the steel  
117 plates screwed to the testing system (Fig. 1). To ensure homogeneity of the stress state in the  
118 specimen, good alignment of the specimen and steel plates is required. The specimen was  
119 conditioned at 20 °C for 4 hours and then tested.

120 During the test, the cyclic torque test returns a dataset which reports the evolution of  
121 torque, axial load, torsional rotation angle and phase angle as a function of the number of cycles.

122 Fig. 2 presents an example of experimental data for two consecutive cycles of loading at 10 Hz.  
 123 Data acquisition frequency is 1,000 Hz, which gives 100 data points per cycle. Because of the  
 124 great number of performed cycles, only one cycle out of 50 has been recorded.



125  
 126

Fig. 2. Experimental data for two loading cycles at 20 °C and 10 Hz.

127 **4. Modelling of the interface behaviour during shear-torque fatigue tests**

128 In this phase, the evolution model of the interlayer complex shear modulus was obtained through  
 129 theoretical consideration involving shear-torque fatigue test results.

130 **4.1 Single-layered specimens**

131 By carrying out monotonic torque tests on cylindrical specimens, it is possible to generate shear  
 132 stresses that linearly increases over the radius of the specimen, as shown in Fig. 3, where  $\delta(t)$   
 133 and  $\gamma(t)$  are, respectively, the circumferential displacement and the shear strain over time.

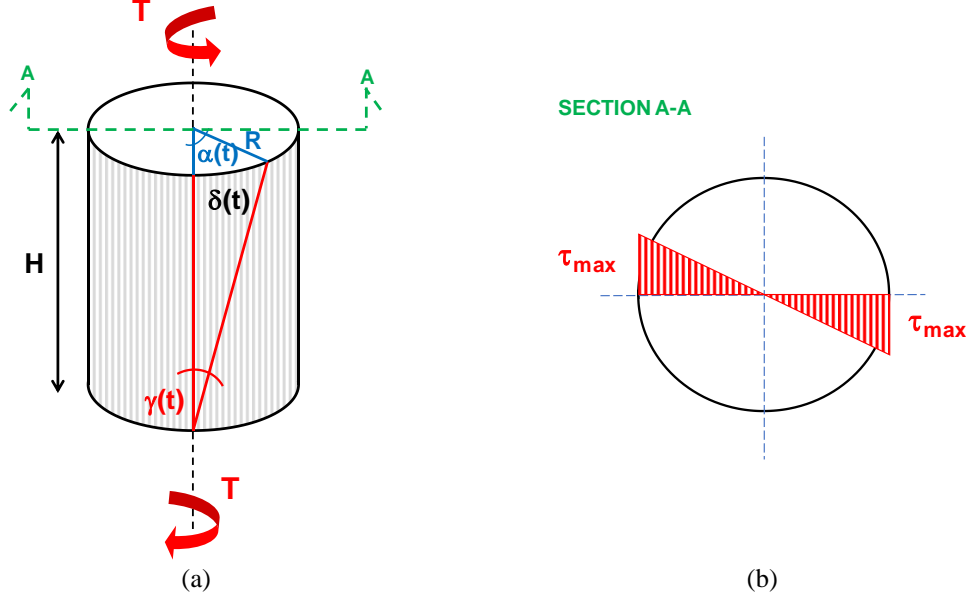
134 The value of the maximum shear stress  $\tau_{max}$  is reached on the outer circumference and can be  
 135 calculated through the following equation:

$$\tau_{max} = \frac{2T}{\pi R^3} \quad (1)$$

136 where  $T$  is the applied torque and  $R$  is the radius of the specimen. Analogously, the maximum  
 137 shear strain  $\gamma_{max}$ , generated by the applied torque, can be calculated as follows:

$$\gamma_{max} = \frac{\delta_{max}}{H} = \frac{R\alpha_{max}}{H} \quad (2)$$

138 where  $H$  is the specimen height and  $\alpha_{max}$  is the torsional rotation angle.



139 Fig. 3. Schematic view of the sample (a) and its section A-A (b) subjected to shear-torque load.

140 When a sinusoidal loading is considered, the maximum shear stress  $\tau_{max}(t)$  can be calculated  
 141 as follows:

$$\tau_{max}(t) = \tau_{max,0} \sin(\omega t) = \frac{2T_0}{\pi R^3} \sin(\omega t) \quad (3)$$

142 where  $T_0$  is the amplitude of the applied torque,  $\omega$  is the torque pulsation ( $\omega = 2\pi f$  with  $f$  the  
 143 load frequency), and  $t$  is the time. The corresponding maximum shear strain  $\gamma_{max}(t)$  can be  
 144 obtained by using Eq. (4):

$$\gamma_{max}(t) = \gamma_{max,0} \sin(\omega t - \varphi) = \frac{R\alpha_0}{H} \sin(\omega t - \varphi) \quad (4)$$

145 where  $\alpha_0$  is the amplitude of the torsional rotation angle (Fig. 2) and  $\varphi$  is the phase angle, which  
 146 is related to the lag between stress and strain.

147 The material stiffness properties have been evaluated in terms of complex shear modulus  
 148  $G^*$  of a single-layered specimen, given by the following equation:

$$G^* = \frac{\tau_{max,0} \exp[j\omega t]}{\gamma_{max,0} \exp[j(\omega t - \varphi)]} = |G^*| \exp[j\varphi] \quad (5)$$

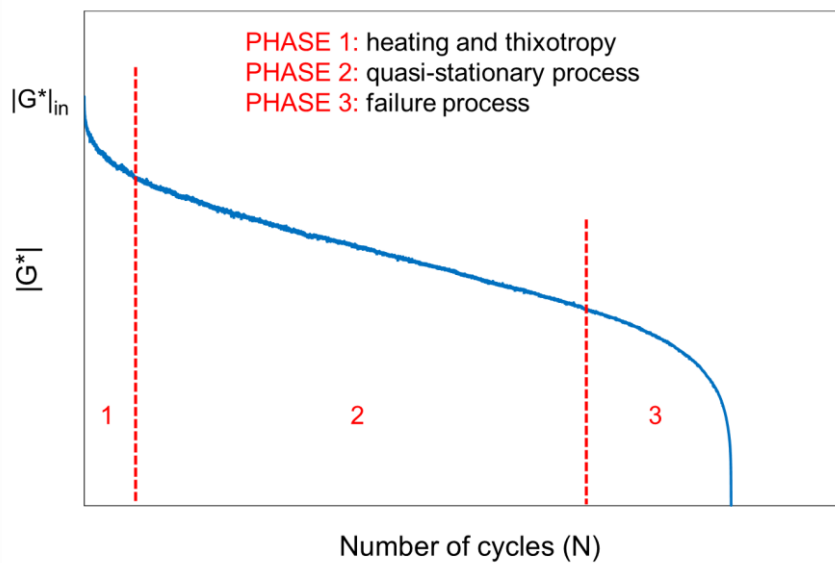
149 where  $j$  is the imaginary unit (defined by  $j^2 = -1$ ) and  $|G^*|$  is the norm (or absolute value) of the  
150 complex shear modulus.

151 By substituting Eqs. (3) and (4) into Eq. (5), the following equation can be obtained:

$$G^* = \frac{H}{I_p} \frac{T_0 \exp[j\omega t]}{\alpha_0 \exp[j(\omega t - \varphi)]} \quad (6)$$

152 where  $I_p = \frac{\pi R^4}{2}$  is the polar moment of inertia of the circular section.

153 Fig. 4 shows a typical example of the evolution of the norm of the complex shear modulus  
154  $|G^*|$  as a function of the number of cycles. During a fatigue test, three phases can be identified  
155 [24,25]. The first phase (or adaption phase), consisting of a rapid decrease of the complex  
156 modulus, is not attributable to fatigue but is caused by bulk reversible phenomena (e.g., self-  
157 heating, thixotropy). The second phase (or quasi-stationary phase) is associated with a quasi-  
158 linear decrease of the complex modulus, during which the initiation and propagation of micro-  
159 cracks within the specimen occur. In the third phase (or failure phase), irreversible phenomena  
160 (e.g., fatigue damage) start and a rapid decrease of the complex modulus begins. During this  
161 phase, macrocracks appear and propagate within the specimen, generating a not homogeneous  
162 distribution of stresses and strains. These three phases are not separated by a clear border but by  
163 a smooth transition zone.



164  
165

Fig. 4. Evolution of the norm of the complex shear modulus as a function of the number of cycles.

166 **4.2 Double-layered specimens**

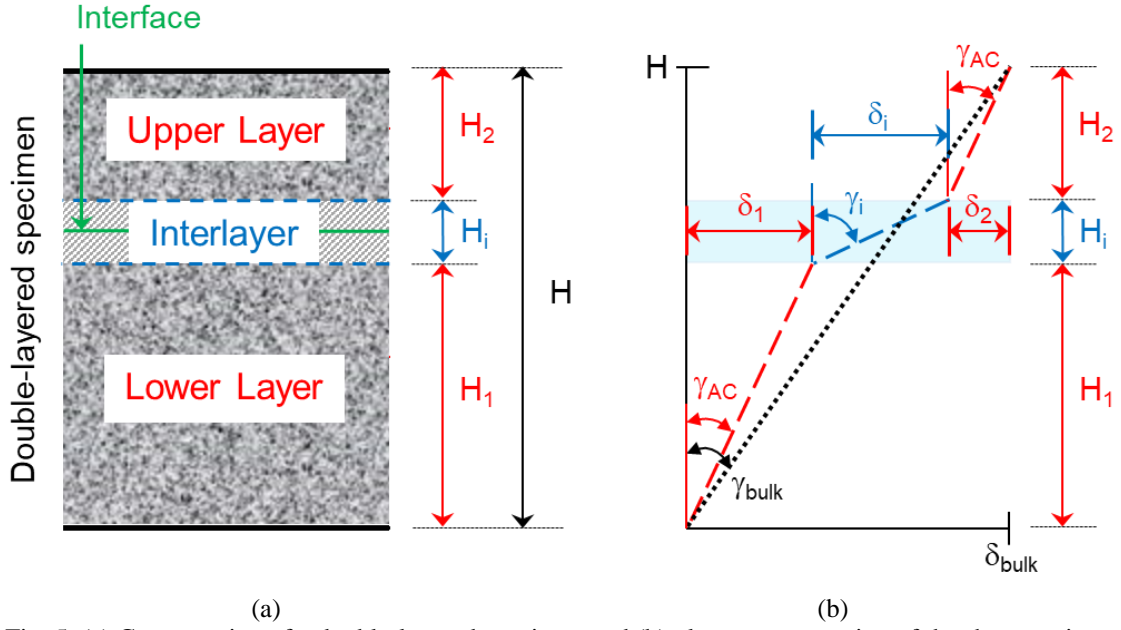
167 The complex shear modulus  $G^*$  can be calculated with Eq. (6) only when single-layered systems  
168 are considered, whereas for double-layered cylindrical specimens the presence of the interface  
169 increases the complexity of the data analysis.

170 Specifically, when double-layered specimens are tested,  $G^*$  can be considered as an  
171 apparent modulus that takes into account the contribution of the two-layer asphalt concrete as  
172 well as the interlayer, assumed as a layer at the interface of appropriate thickness. Hence, the total  
173 height of the specimen ( $H$ ) becomes equal to the sum of the heights of the lower layer ( $H_1$ ), the  
174 upper layer ( $H_2$ ) and the interlayer ( $H_i$ ) (Fig. 5.a). Analogously, the amplitude of the torsional  
175 rotation angle  $\alpha_0$  comprises the torsional rotation angle of the asphalt mixture that characterises  
176 the lower and the upper layers plus the torsional rotation angle of the interlayer.

177 Since the fracture usually occurs in correspondence of the interface [23], it can be  
178 assumed that the shear strain of the interlayer  $\gamma_i$  and the interlayer complex shear modulus  $G_i^*$  are  
179 strictly linked to the specimen failure. Then, these two parameters were analytically determined  
180 in order to evaluate their evolution until failure.

181 In the following equations, the parameters obtained during the double-layered shear-  
182 torque fatigue tests and referred to the overall response of the specimen are named with the  
183 subscript “*bulk*”, the parameters referred to the two layers in asphalt concrete are named with the  
184 subscript “*AC*” and the parameters referred to the interlayer are named with the subscript “*i*”.

185 It can be assumed that the general trend of the maximum shear strain  $\gamma_{bulk}$  (i.e.,  $\gamma_{max}$ ),  
186 obtained during the double-layered test and denoted with a black dotted line in Fig. 5.b, can be  
187 schematically represented by the contribution of the shear strain of the asphalt concrete  $\gamma_{AC}$  (red  
188 dashed lines) and by the contribution of the interlayer shear strain  $\gamma_i$  (blue dashed line).



189 Fig. 5. (a) Cross-section of a double-layered specimen and (b) plane representation of the shear strain  
 190 interpretation (the figure is not drawn to scale).

191 Based on this assumption, it is possible to separate the different contributions to the shear strain.

192 Specifically, the value of the interlayer shear strain  $\gamma_i$  can be calculated as follows:

$$\gamma_i = \frac{\delta_i}{H_i} \quad (7)$$

193 where  $\delta_i$  can be obtained by the following equation:

$$\delta_i = \delta_{bulk} - (\delta_1 + \delta_2) \quad (8)$$

194 Assuming that both layers are prepared with the same asphalt concrete, it can be considered that

195  $\delta_1 = \gamma_{AC} \cdot H_1$  and  $\delta_2 = \gamma_{AC} \cdot H_2$ , leading the following equation:

$$\delta_i = \delta_{bulk} - \gamma_{AC} \cdot (H_1 + H_2) = \delta_{bulk} - \gamma_{AC} \cdot (H - H_i) \quad (9)$$

196 The shear strain of the asphalt concrete  $\gamma_{AC}$  can be calculated from the norm of the complex shear

197 modulus  $|G_{AC}^*|$ , as follows:

$$\gamma_{AC} = \frac{\tau_{max,0}}{|G_{AC}^*|} \quad (10)$$

198 Considering that  $\delta_{bulk} = R\alpha_0$ , the substitution of Eqs. (9) and (10) in Eq. (7), allows to obtain:

$$\gamma_i = \frac{R\alpha_0 - \frac{\tau_{max,0}}{|G_{AC}^*|} \cdot (H - H_i)}{H_i} \quad (11)$$

199 Since the interlayer complex shear modulus  $G_i^*$  is expressed by the following equation:

$$G_i^* = \frac{\tau_{max,0} \exp[j\omega t]}{\gamma_i \exp[j(\omega t - \varphi)]} \quad (12)$$

200 its norm can be calculated, by using Eq. (11), as follows:

$$|G_i^*| = \frac{\tau_{max,0}}{\gamma_i} = \frac{\tau_{max,0} \cdot H_i}{R\alpha_0 - \frac{\tau_{max,0}}{|G_{AC}^*|} \cdot (H - H_i)} \quad (13)$$

201 Thus, the determination of the norm of the interlayer complex shear modulus  $|G_i^*|$  is subjected to  
 202 the knowledge of the interlayer thickness  $H_i$  and of the norm of the complex shear modulus of  
 203 the asphalt concrete  $|G_{AC}^*|$ . The influence of both parameters ( $|G_{AC}^*|$  and  $H_i$ ) on the application of  
 204 Eq. (13) are studied in depth in Section 6.

## 205 **5. Experimental investigation**

### 206 **5.1 Materials and specimen preparation**

207 The asphalt concrete (AC) used for this study was a dense-graded mixture for surface courses  
 208 with 12.5 mm nominal maximum aggregate size (Fig. 6). The AC was prepared with a polymer-  
 209 modified bitumen (PMB) classified as 45/80-65 (EN 12591 [26]) and dosed at 5.5% by aggregate  
 210 mass.

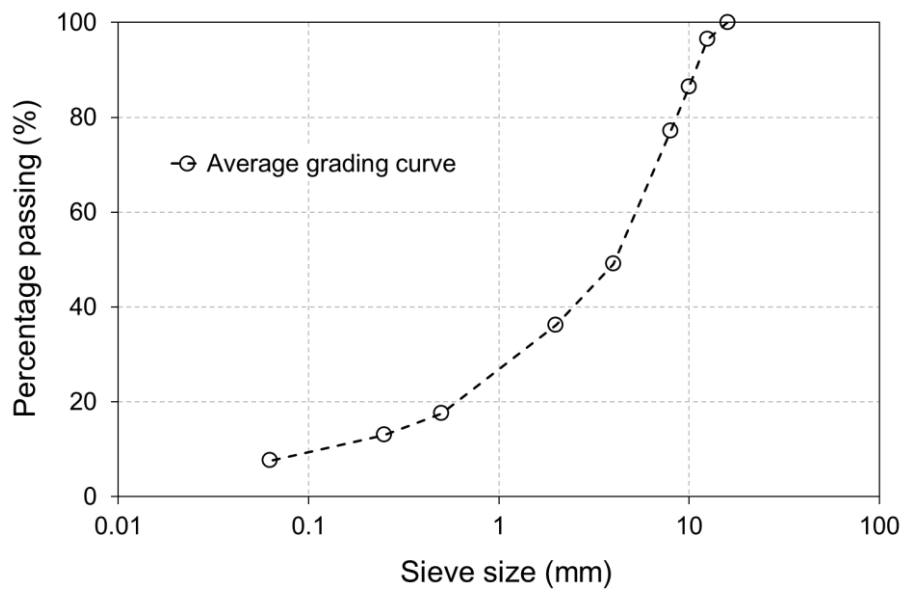


Fig. 6. Aggregate grading curve of the AC mixture.

211

212

213 The AC was compacted in the laboratory by means of a steel roller compactor (EN 12697-33  
 214 [27]), manufacturing three slabs ( $305 \times 305 \times 100 \text{ mm}^3$ ): one single-layered, named as A, and  
 215 two double-layered, named as B and C. The slab A had a thickness of 100 mm, whereas the lower  
 216 and the upper layers of slabs B and C had a thickness of 60 mm and 40 mm, respectively. No  
 217 tack-coat was spread at the interface. All the layers were characterised by the same target air voids  
 218 content (5%), to allow a reliable comparison in terms of interface performance.

219 Five cylindrical specimens (with a nominal diameter  $D = 100 \text{ mm}$ ) were cored from each  
 220 slab (Fig. 7). Once cored, the upper and the lower parts of each specimen were sawed (about  
 221 10 mm) and, subsequently, trimmed with an abrasive disc circular saw in order to provide a ridge-  
 222 free surface and to ensure parallel alignment of the edges of the specimen (Fig. 8). Only one  
 223 cylindrical specimen without interface (obtained from slab A) and ten double-layered specimens  
 224 (obtained from slabs B and C) were tested and analysed in the second and third phase of this  
 225 investigation.

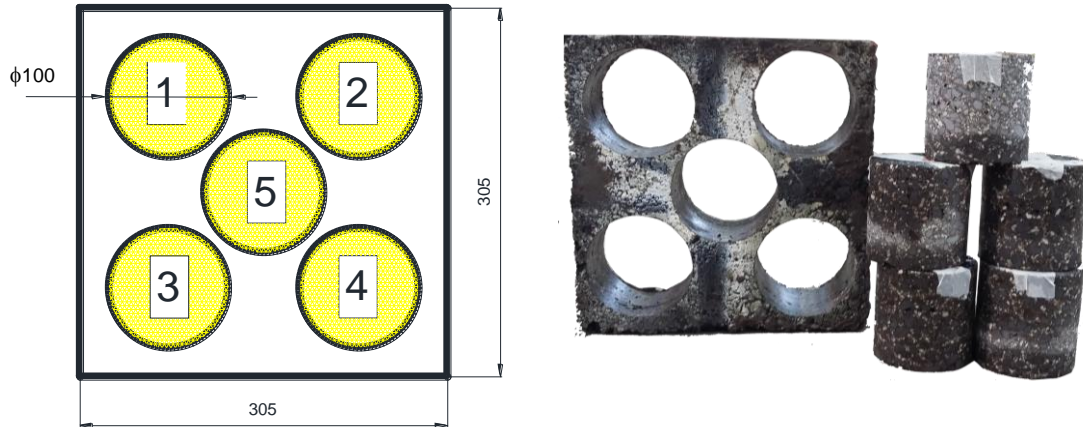


Fig. 7. Cylindrical specimens cored from the slabs.

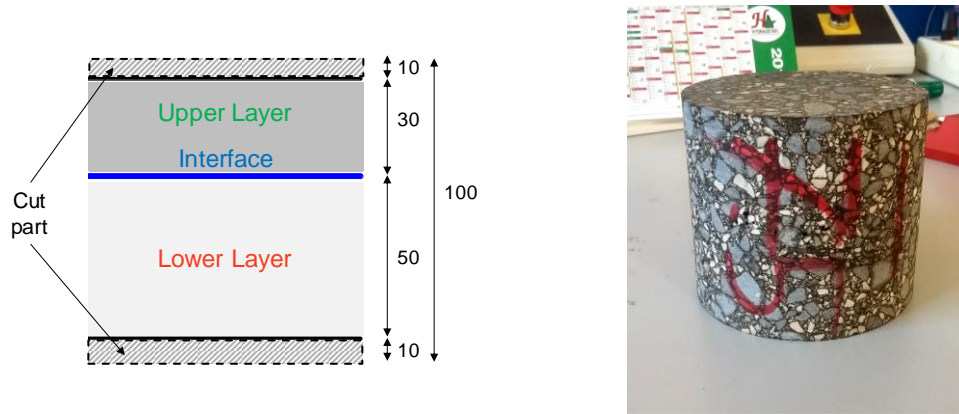


Fig. 8. Sawed and trimmed cylindrical specimens of slabs B and C.

226 **5.2 Laboratory program**

227 All the shear-torque fatigue tests were performed according to the procedure described in  
 228 Section 3. Constant torque amplitudes ( $T_0$ ) ranging from 70 to 100 Nm were applied during the  
 229 investigation, according to a preliminary study which showed that these stress levels guarantee a  
 230 proper number of cycles before failure.

231 For the preliminary validation of the model, shear-torque fatigue tests were performed on  
 232 both one specimen of slab A (without interface) and one specimen of slab C (with interface),  
 233 applying a torque amplitude  $T_0$  equal to 80 Nm.

234 Finally, in order to correctly characterise the fatigue interlayer behaviour through the  
 235 determination of the fatigue curve, the testing program shown in Table 1 was carried out. One

236 different level of torque amplitude was applied to each specimen, by stopping the test when the  
 237 complete fracture of the specimen was obtained (i.e., the separation between the two layers).

238 Each specimen was named by means of an identification code (ID) that indicates the slab,  
 239 the coring position (Fig. 7) and the applied torque amplitude; for example, A\_2\_80 represents the  
 240 test on the specimen cored in the slab A, in the position #2, performed with a torque amplitude  
 241 equal to 80 Nm.

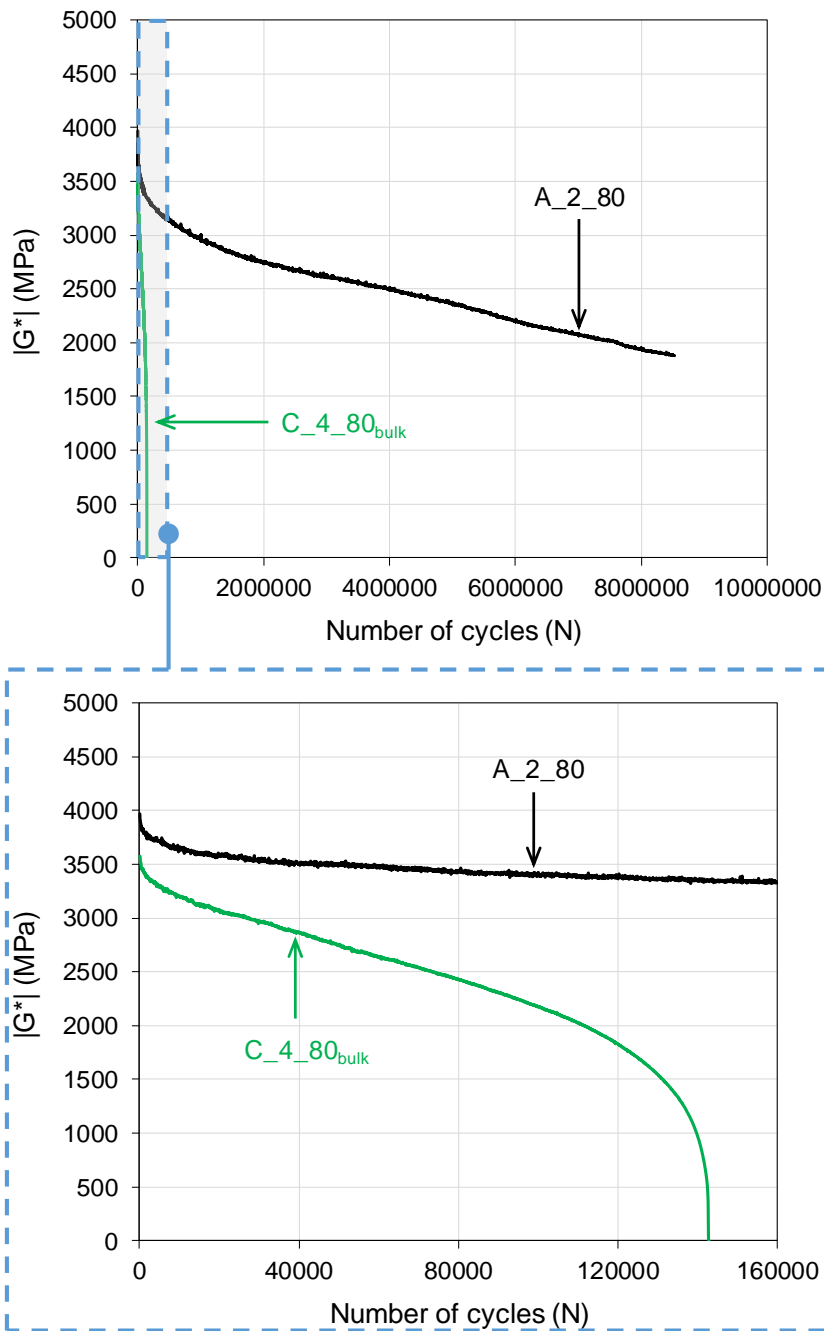
242 Table 1. Experimental program of the third phase.

Slab	Specimen position #	Temperature °C	Frequency Hz	Torque amplitude $T_0$ Nm	Specimen ID
	1			100	B_1_100
B	2	20	10	95	B_2_95
	5			85	B_5_85
-----					
	1			70	C_1_70
	2			85	C_2_85
C	3	20	10	75	C_3_75
	4			80	C_4_80
	5			90	C_5_90

243 **6. Preliminary validation of the model**

244 The values of the norm of the complex shear modulus of A\_2\_80 (without interface) and C\_4\_80  
 245 (with interface) are shown in Fig. 9, as a function of the number of loading cycles.

246 Since the specimens were characterised by an analogous bulk density (about 2.42 g/cm<sup>3</sup>)  
 247 and by the same torque amplitude ( $T_0 = 80$  Nm), Fig. 9 provides evidence that the interface  
 248 represents a weakness zone, leading to early failure compared to the specimen without interface.  
 249 Indeed, specimen C\_4\_80 failed after about 143,000 cycles with an evolution of  $|G^*|$  showing the  
 250 three phases of the decay law seen in Fig. 4, whereas the specimen A\_2\_80 did not fail even after  
 251 8,500,000 cycles. Due to the excessive testing time, the A\_2\_80 test was stopped when the  
 252 decrease of the norm of the complex shear modulus was approximately 58%, before reaching the  
 253 failure condition. For this reason, the first and the second phase of the decay law can be identified,  
 254 whereas the third phase cannot be seen.



255

256

Fig. 9. Evolution of  $|G^*|$  for specimens A\_2\_80 and C\_4\_80<sub>bulk</sub> during shear-torque fatigue test.

257

The remarkable difference just observed between the single-layered and the double-layered

258

specimen confirms the assumption that the interlayer shear strain  $\gamma_i$  controls the failure of the

259

latter. Thus, the effect of the presence of the interface on the response of double-layered systems

260

under shear-torque fatigue loading was assessed by considering the interlayer complex shear

261

modulus  $|G_i^*|$  of the specimen C\_4\_80, determined according to Eq. (13). However, as above-

262 mentioned, a proper estimation of the norm of the complex shear modulus  $|G_{AC}^*|$  and of the  
263 interlayer thickness  $H_i$  is fundamental for a correct evaluation of  $|G_i^*|$ . In the following  
264 observations, the variation of each of the two parameters was considered separately.

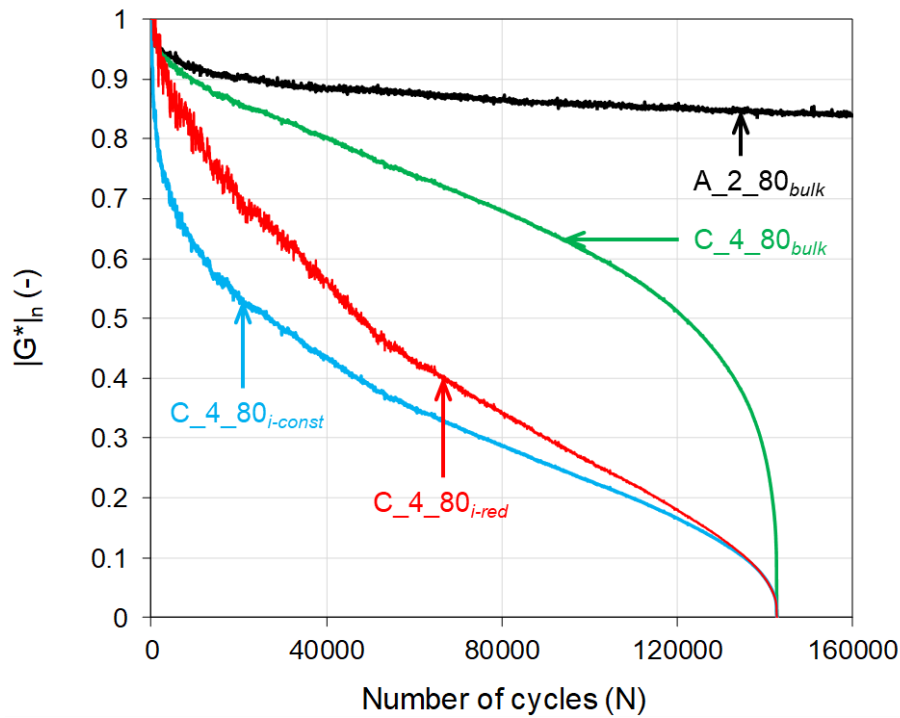
265 As regards the first parameter, it is necessary to determine if  $|G_{AC}^*|$  can be assumed  
266 constant during the whole double-layered specimen test or if it must be considered progressively  
267 reduced. Indeed, during the test,  $|G_{AC}^*|$  decreases due to the specimen damaging and it is important  
268 to understand the influence of this reduction on the  $|G_i^*|$  determination. To this end, both a  
269 constant value and a progressive reduction law of  $|G_{AC}^*|$  were considered in Eq. (13), by fixing a  
270 realistic interlayer thickness of 15 mm, according to Ragni et al. findings [23].

271 As far as the constant value law of  $|G_{AC}^*|$  is concerned, it can be assumed that the double-layered  
272 specimen initially reacts as a single-layered system, so that the  $|G_{AC}^*|$  can be considered equal to  
273 the  $|G^*|_{bulk}$  of the C\_4\_80 in the first loading cycles (when the specimen is not yet damaged),  
274 allowing the determination of  $|G_i^*|_{const}$  by means of Eq. (13).

275 On the other hand, the progressive reduction of  $|G_{AC}^*|$  was evaluated through the study of the  
276 behaviour of the single-layered specimen A\_2\_80. Specifically, Fig. 9 clearly shows that, during  
277 the testing time of C\_4\_80 (about 143,000 cycles), also  $|G^*|$  of A\_2\_80 is subjected to a gradual  
278 reduction, leading to consider that also the AC outside the interlayer likely suffers some damage  
279 during the C\_4\_80 shear-torque fatigue loading. Thus, cycle by cycle, the decay law of  $|G^*|$  of  
280 A\_2\_80 was deduced and used for obtaining the decay law of  $|G_{AC}^*|$  of C\_4\_80 (named  $|G_{AC}^*|_{red}$ ).  
281 This law was introduced into Eq. (13), by allowing the calculation of  $|G_i^*|_{red}$  of C\_4\_80 in  
282 correspondence of each cycle.

283 Finally, in order to allow an easier comparison, the values assumed by  $|G_i^*|_{const}$  and  $|G_i^*|_{red}$  at  
284 the  $n^{th}$  cycle were normalised with respect to their initial values and plotted as a function of the  
285 number of loading cycles, as shown in Fig. 10 (C\_4\_80<sub>i-const</sub> and C\_4\_80<sub>i-red</sub>, respectively). In the  
286 same figure, also the evolution of the normalised moduli of the specimen without interface,  
287 A\_2\_80, and of the double-layered specimen C\_4\_80<sub>bulk</sub> are shown.

288 The differences observed between  $C\_4\_80_{i-const}$  and  $C\_4\_80_{i-red}$  in the greatest part of the test are  
 289 obviously due to the fact that in the calculation of  $C\_4\_80_{i-const}$  it is assumed that all the damage  
 290 is immediately localised at the interface, whereas, for  $C\_4\_80_{i-red}$ , part of the damage affects also  
 291 the AC outside the interlayer. However, in the proximity of the end of the test, the two interlayer  
 292 moduli ( $|G_i^*|_{const}$  and  $|G_i^*|_{red}$ ) become very similar and thus, at this stage, the effect of the  
 293 reduction of  $|G_{AC}^*|$  plays a minor role in the damage trend of  $|G_i^*|$ . Thus, since it has been shown  
 294 by Ragni et al. [23] that the “effective” damage at the interface occurs in the proximity of the end  
 295 of the test, the assumption of considering a constant value for  $|G_{AC}^*|$  can be considered valid  
 296 without committing significant errors.

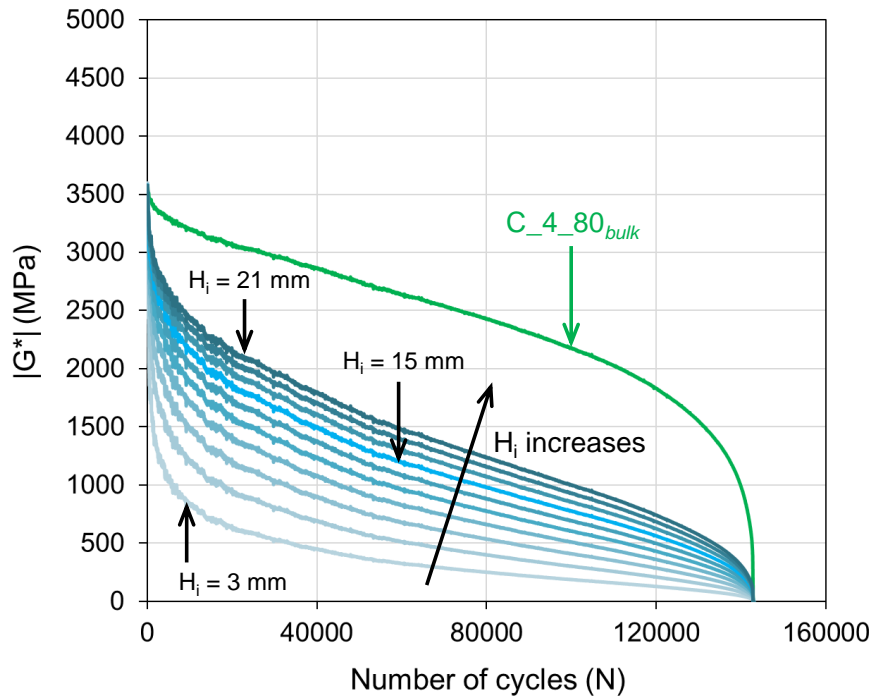


297  
 298 Fig. 10. Evolution of  $|G^*|_n$  for specimens A\_2\_80 and C\_4\_80\_bulk during shear-torque fatigue test.

299 The second aspect, that should be taken into account, is the selection of the proper value for the  
 300 interlayer thickness  $H_i$ . In this regard, Ktari et al. [28] performed monotonic tensile tests on  
 301 double-layered asphalt specimens and used the Digital Image Correlation (DIC) technique to  
 302 identify the interphase thickness (i.e., interlayer thickness  $H_i$ ). They found out that the  $H_i$  can be  
 303 assumed equal to 3 mm. In this study, a range of variation from 3 to 21 mm, with a step of 2 mm,

304 was considered for this parameter ( $H_i$ ), by assuming a constant value for  $|G_{AC}^*|$ , equal to  $|G^*|_{bulk}$   
 305 of C\_4\_80 in the first loading cycles (see above).

306 Fig. 11 presents  $|G_i^*|$  (in blue), obtained with different values of  $H_i$ , as a function of the  
 307 number of loading cycles, as well as the comparison with the corresponding  $|G^*|_{bulk}$  (named as  
 308 C\_4\_80<sub>bulk</sub>). Fig. 11 clearly shows that  $|G_i^*|$  curves are characterised by the typical three phases  
 309 of fatigue tests. At the beginning,  $|G_i^*|$  decreases drastically and then decreases quite steadily until  
 310 approaching the end of the test, when a sudden reduction of the interlayer modulus occurs. It can  
 311 be observed that a sudden decrease in the initial value of  $|G_i^*|$  occurs for low  $H_i$  values (i.e., less  
 312 than 11 mm), indicating that  $|G_i^*|$  is markedly affected by the value of the interlayer thickness  $H_i$ .  
 313 The higher the  $H_i$  value, the closer  $|G_i^*|$  to the behaviour of  $|G^*|_{bulk}$  (C\_4\_80<sub>bulk</sub>).

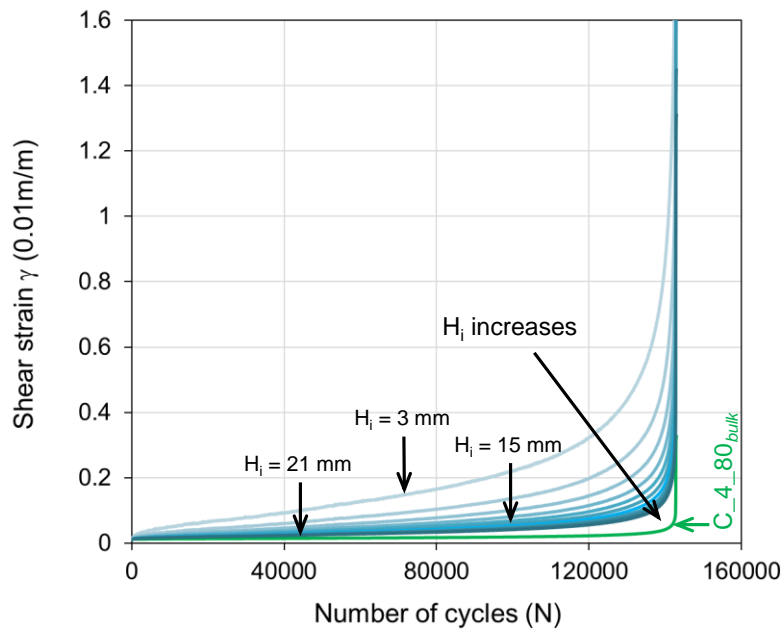


314

315 Fig. 11. Evolution of  $|G^*|_{bulk}$  and  $|G_i^*|$  for specimen C\_4\_80<sub>bulk</sub> during shear-torque fatigue test.

316 Since the applied torque induces a mechanical response within the material in the form of shear  
 317 strain  $\gamma$  and the increase in the strain is usually due to damage accumulation in the material, it is  
 318 interesting to evaluate the influence of the interlayer thickness  $H_i$  on this parameter. In Fig. 12,  
 319 the shear strain of the bulk system  $\gamma_{bulk}$  and the interlayer shear strain  $\gamma_i$  of the specimen C\_4\_80

320 (in blue) for different values of  $H_i$ , are reported. The shear strain of the bulk system  $\gamma_{bulk}$  shows  
 321 a two-phase process consisting of an almost constant trend until the proximity of the end of the  
 322 test is reached when a sharp increase occurs. It is therefore clear that the double-layered specimen  
 323 absorbs the shear-torque fatigue loading by producing very low shear deformations for the  
 324 greatest part of the test duration. On the contrary,  $\gamma_i$  shows three distinct phases, whose evolution  
 325 strongly depends on the value of  $H_i$ . The higher the value of  $H_i$ , the closer  $\gamma_i$  to the behaviour of  
 326  $\gamma_{bulk}$ . Specifically, changes in shear strain appear to be more noticeable assuming an interlayer  
 327 thickness less than 11 mm. These outcomes suggest that  $H_i$  appears to be the variable which plays  
 328 the most important role in the interlayer characterisation. For this reason, further research focused  
 329 on the accurate identification of  $H_i$  (for example, by utilising digital image correlation technique)  
 330 can considerably help in the development of a more precise evaluation of the interlayer behaviour.  
 331 Nevertheless, an interlayer thickness  $H_i$  equal to 15 mm was assumed in the following discussion,  
 332 according to Ragni et al. [23] findings.



333  
 334 Fig. 12. Evolution of  $\gamma_{bulk}$  and  $\gamma_i$  for specimen C\_4\_80\_bulk during shear-torque fatigue test.

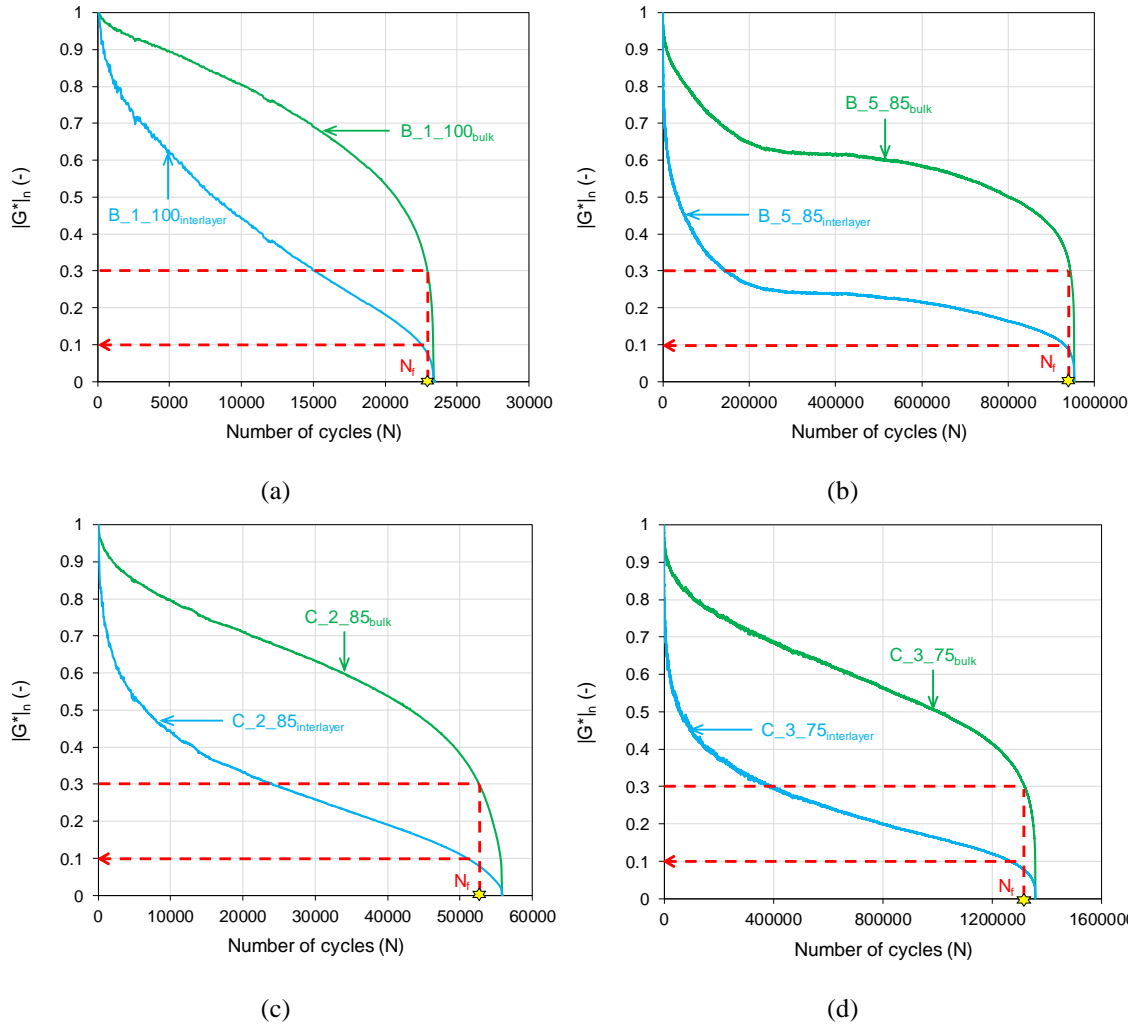
335

## 336 7. Interlayer shear fatigue curve

### 337 7.1 Fatigue failure criterion

338 A proper failure criterion should predict the fatigue life of the material and capture the failure of  
339 the specimen accurately (i.e., in this case, the formation of macro-cracks at the interface). In this  
340 regard, Ragni et al. [23] showed that, for shear-torque fatigue tests performed on double-layered  
341 specimens, the classical fatigue failure criterion (which defines failure as the point at which the  
342 stiffness modulus value reduces to 50% of its initial value) seems to provide an inaccurate  
343 estimation of the number of cycle to failure ( $N_f$ ) since it is not able to quantify the damage  
344 mechanisms that occur within double-layered specimens. In the same study [23], the authors  
345 suggested that a failure criterion characterised by a 70% reduction of  $|G^*|_{bulk}$  seems to better  
346 describe the material behaviour under shear-torque fatigue loading.

347 As an example, Fig. 13.a through Fig. 13.d present the criterion of the 70% reduction of  
348  $|G^*|_{bulk}$  applied to four test results in this study. These graphs suggest that the number of cycles  
349 to failure ( $N_f$ ) determined based on this failure criterion correspond to a reduction of the interlayer  
350 complex shear modulus  $|G_i^*|$  approximately equal to 90%, that is  $|G^*|_n = 0.1$ . Consequently,  
351 rather than to use the 70% reduction of  $|G^*|_{bulk}$ , the 90% reduction of the initial value of the  
352 interlayer complex shear modulus  $|G_i^*|$  can be adopted as a failure criterion for shear-torque  
353 fatigue test carried out on double-layered asphalt specimens. Moreover, it can also be noted that  
354 the evolution of  $|G^*|_n$  for the specimen B\_5\_85 (Fig.13.b) significantly differs from that of the  
355 other specimens. An analogous trend was also found for the specimen C\_5\_90, not reported in  
356 Fig. 13. This behaviour could be due to the crack pattern at the interface obtained for these two  
357 specimens, as described in the following section.



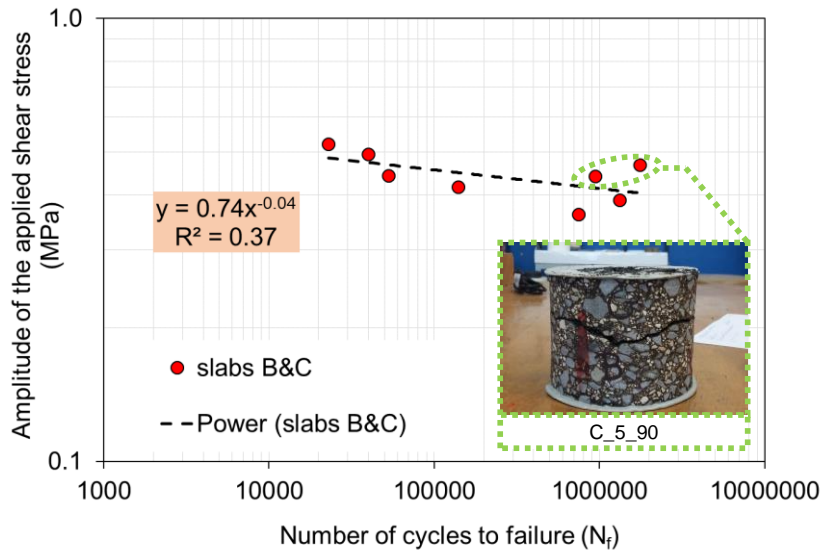
358 Fig. 13. Evolution of  $|G^*|_n$  during shear-torque fatigue test and fatigue failure criterion: (a) B\_1\_100; (b)  
 359 B\_5\_85; (c) C\_2\_85; (d) C\_3\_75.

## 360 7.2 Fatigue curve

361 In Fig. 14, the interlayer shear fatigue curve of the specimens cored from slabs B and C is plotted  
 362 in a log-log plane. Such analysis was based on a power-law relationship between the amplitude  
 363 of the applied shear stress  $\tau_{max,0}$  and the number of loading cycles to failure  $N_f$ , recorded in  
 364 correspondence of the 90% reduction of the initial value of  $|G_i^*|$ , using the following equation:

$$\tau_{max,0} = a \cdot N_f^{-b} \quad (14)$$

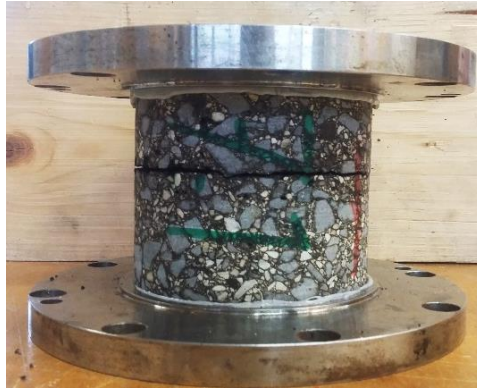
365 where the parameters  $a$  and  $b$  are regression coefficients and usually depend on both material and  
 366 testing temperature. Specifically,  $b$  represents the slope of the linear regression which is obtained  
 367 in a log-log plane.



368 Fig. 14. Interlayer shear fatigue curve in terms of the amplitude of the applied shear stress as a function of  
 369 the number of cycles to failure.  
 370

371 Fig. 14 shows that, in general, as the level of shear stress (i.e., torque) decreases, the shear fatigue  
 372 life of the bituminous material increases even if the correlation between the obtained data is not  
 373 good ( $R^2 = 0.37$ ). This can be due to the fact that the two data points circled in green, do not  
 374 follow the general trend as they provide higher values of  $N_f$  for higher values of  $\tau_{max,0}$ . Looking  
 375 for the reason for this behaviour, it was observed that, in both cases (specimens B\_5\_85 and  
 376 C\_5\_90), the crack pattern at the interface was irregular (see the box in Fig. 14), contrarily to  
 377 what happened for all the other specimens, for which the fracture occurred exactly at the interface  
 378 (Fig. 15). This finding corroborates what observed in Fig. 13, where different evolution of  $|G^*|_n$   
 379 were found for B\_5\_85 (as well as C\_5\_90) with respect to the others. As above-mentioned, this  
 380 different behaviour can be due to the different crack pattern at the interface observed for these  
 381 two specimens, suggesting that a visual failure inspection of the specimens is essential for  
 382 developing a correct interlayer fatigue characterisation. Moreover, it was noticed that both these  
 383 specimens (B\_5\_85 and C\_5\_90) were cored at the centre of the slabs (i.e., position #5 of Fig. 7),  
 384 where a better interlocking between the two bituminous layers in contact could be achieved due  
 385 to the not homogeneous compaction process performed by the roller compactor (i.e., higher  
 386 compaction energy is expected in the centre of the slab) [29,30]. However, it should be mentioned

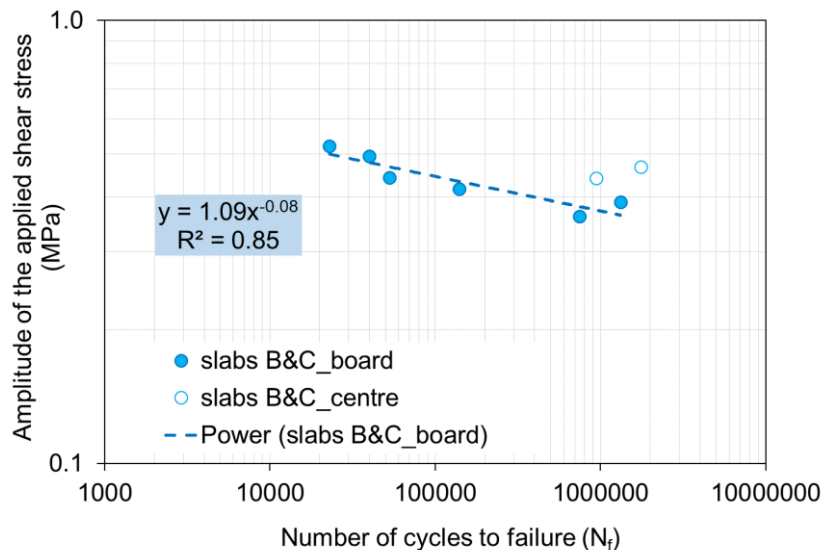
387 that, among the laboratory methods used to fabricate laboratory specimens, the roller compactor  
388 provides the best correlation with field cores in terms of both internal aggregate structure and  
389 mechanical properties [31-33].



390

391 Fig. 15. Failure mechanisms of the specimens at the border of the slab.

392 The updated interlayer shear fatigue curve is reported in Fig. 16, in which the specimens cored in  
393 position #5 (empty dots) were not considered. In this case, the good correlation coefficient  
394 ( $R^2 = 0.85$ ) clearly indicates that the coring position is relevant, at least for shear-torque fatigue  
395 tests.



396

397 Fig. 16. Interlayer shear fatigue curve of the border specimens in terms of the amplitude of the applied  
398 shear stress as a function of the number of cycles to failure.

399 It can be concluded that the fatigue process of double-layered specimens is a complex  
400 phenomenon due to a twofold factor: the heterogeneous nature of the bituminous material and the

401 presence of a weakness zone (i.e., interlayer). Other viscoelastic phenomena occur during fatigue  
402 tests (e.g., self-heating, thixotropy, non-linearity) but have not been taken into account in this  
403 study.

## 404 **8. Conclusions**

405 This paper focuses on the evaluation of the fatigue behaviour of pavement bituminous interlayers  
406 subjected to shear-torque fatigue tests. A preliminary experimental investigation was carried out  
407 on single- and double-layered specimens to investigate the influence of the interface on shear-  
408 torque fatigue properties. Based on these preliminary finding, a model describing the evolution  
409 of the interlayer damage and a new fatigue failure criterion were proposed in order to correctly  
410 identify the material behaviour at the interface and to overcome the classical damage approach.

411 The following main conclusions can be drawn:

- 412 - the presence of the interface represents a weakness zone in the double-layered specimen  
413 and, therefore, it noticeably influences the shear fatigue performance;
- 414 - the norm of the interlayer complex shear modulus and the norm of the complex shear  
415 modulus decrease as the number of loading cycles increases, indicating that the first can  
416 be a good indicator of the damage of the specimen interface;
- 417 - the proposed model describes the evolution of the interlayer damage of double-layered  
418 specimens under shear-torque fatigue tests through the analysis of the variation of the  
419 norm of the interlayer complex shear modulus. Preliminary validation of the model  
420 showed that it adequately characterises the evolution of the interlayer properties during  
421 cyclic torque loading even if further refinements to the model are needed;
- 422 - a new fatigue failure criterion, which seems to better correspond to the mechanisms of  
423 failure that occurred at the interface of double-layered specimens, was proposed for shear-  
424 torque fatigue test;

425 - a comparison between specimens cored at the centre and at the border of the compacted  
426 slabs seems to show the existence of non-uniform compaction which is reflected in a  
427 dissimilar shear fatigue life. This result must be confirmed by future experimentation.

428 Summarising, this study suggests that the proposed testing method and analysis can effectively  
429 provide a noticeable help for the understanding of shear-torque fatigue phenomena at the interface  
430 of double-layered asphalt concrete specimens. The work herein represents a fundamental basis  
431 for future developments required to confirm the theoretical assumptions. Further experiments will  
432 be carried out by broadening the tested materials (layers composed of different asphalt mixtures)  
433 to assess the reliability and accuracy of the model as well as to validate the proposed fatigue  
434 failure criterion.

#### 435 **Declaration of competing interest**

436 The authors declare that they have no known competing financial interests or personal  
437 relationships that could have appeared to influence the work reported in this paper.

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